

LAMPIRAN

Kode Program Matlab

```
clear
clc
%
% Program untuk menentukan natural static & rotating mode shapes
% frequencies pada balok yang seragam/uniform menggunakan model UH-60
%
mode=3;
p=100; % banyak elemen
diskritisasi;
alt=0; % bisa 0, 5k, 10k
cgea=0.25; %input(5);
mn=1; % input('Vfwd
minimum? ');
mx=2000; % input('Vfwd
maximum? ');
%
R=322; offset=15; chord=20.76; % in
L=R-offset; e=offset/L; % in
O=27.02; % kecepatan rotor
rad/s
mu=0.00164167; % lbs^2/in^2
E=10000000; % lb/in^2
EIxx=2.27859e7; EIyy=7.71261e8; % lbin^2
I=(EIxx+EIyy)/E; % in^4
Ia=0.037006386; % lbins^2/in
polar mass MOI trhadap e.a.
GJ=2.467909e7; % lbin^2
%
a=-0.5; % non-dim elastic
axis location measured from midchord
b=chord/2; % panjang
setengah chord dlm m
if alt==10
    rho=0.0017556/(12^4); % 10000 ft %
    lbs^2/in^4 density of air
elseif alt==5
    rho=0.0020482/(12^4); % 5000 ft
else
    rho=0.0023769/(12^4); % Sea Level
end
%
d=R/p; % panjang tiap
satu elemen
%
%natural frequency
asubn=[15.41820562 49.96486209 104.2476966]; % Hartog
K0=[6.38 17.63 35.05]; K1=[9.18 26.02 52.2]; % Yntema
BL=[3.926602 7.068583 10.21018]; % bending fixed-
free
```

```

an=[(sin(BL(:,1))-sinh(BL(:,1)))/(cos(BL(:,1))-cosh(BL(:,1)))
(sin(BL(:,2))-sinh(BL(:,2)))/(cos(BL(:,2))-cosh(BL(:,2)))
(sin(BL(:,3))-sinh(BL(:,3)))/(cos(BL(:,3))-cosh(BL(:,3)))];
pf=cos(BL(:,1))*cosh(BL(:,1)); %persamaan
frekuensi
%
%natural frekuensi tidak memutar
%
wnb=[(BL(:,1))^2*(sqrt((EIxx)/(mu*R^4)))
(BL(:,2))^2*(sqrt((EIxx)/(mu*R^4)))
(BL(:,3))^2*(sqrt((EIxx)/(mu*R^4)))]; %natural frekuensi bending
tidak memutar
wnt=[0.5 1.5 2.5]*pi*sqrt(GJ/(Ia*R^2)); %natural frekuensi
torsion tidak memutar
%
%natural frekuensi memutar
%
wnbp=sqrt(wnb.*wnb+(K0+K1*e).*O^2) ; %bending
wntp=(O.^2+wnt.^2).^^(1/2) ; %torsion
%
r(1)=0; % r is the vector
of element edges in percent R
for q=1:p
    r(q+1)=q/p;
end
for n=1:p
    mid(n)=(r(n+1)+r(n))/2; % titik tengah
tiap elemen
end
%
%menentukan modeshape bending fixed-engsel
%
u=mid'*BL;
for n=1:3
    for m=1:p
        h=u(m,n);
        f(p+1-m,n)=(cosh(h)+cos(h)-an(n)*(sinh(h)+sin(h)))/2;
    end
end
%
for n=1:3 % This section normalizes mode shape
    if f(1,n)<0 % to all begin with a positive slope
        h=-1*f(:,n); % for display purposes
        f(:,n)=h; % (does not affect calculations)
    end
end
for n=1:3 % This section calculates the
torsional mode shapes % According to Hartog (already
    for m=1:p % normalized to 1)
        F(m,n)=sin((n-.5)*pi*mid(m));
    end
end
%
%=====
=====

```

```

%Plot mode shapes
%=====
=====

figure(1)      % Plot of mode shapes
title('Mode Shapes untuk Bending dan Torsi')
plot(mid,f(:,1),mid,f(:,2),mid,f(:,3),mid,F(:,1),mid,F(:,2),mid,F(:,3)
)
grid on
axis([min(mid),max(mid),-1,1])
legend('f(1)', 'f(2)', 'f(3)', 'F(1)', 'F(2)', 'F(3)')
xlabel('normalized panjang beam');ylabel('Normalized Defleksi');

figure(2)      %mode shapes bending
title('Mode Shapes untuk Bending')
plot(mid,f(:,1),mid,f(:,2),mid,f(:,3))
grid on
axis([min(mid),max(mid),-1,1])
legend('f(1)', 'f(2)', 'f(3)')
xlabel('normalized panjang beam');ylabel('Normalized Defleksi
Bending');

figure(3)      %mode shapes torsi
title('Mode Shapes untuk Torsi')
plot(mid,F(:,1),mid,F(:,2),mid,F(:,3))
grid on
axis([min(mid),max(mid),-1,1])
legend('F(1)', 'F(2)', 'F(3)')
xlabel('normalized panjang beam');ylabel('Normalized Defleksi Torsi');
%
%
%normalize bending
fact1=sqrt(f(:,1)'*f(:,1)*mu*d);
fact2=sqrt(f(:,2)'*f(:,2)*mu*d);
fact3=sqrt(f(:,3)'*f(:,3)*mu*d);
wb(:,:)=f(:,:); %/fact1;
mb=[wb(:,1) wb(:,2) wb(:,3)]'*[wb(:,1) wb(:,2) wb(:,3)]*mu*d; %cek
apakah identitas
%
%normalized torsi
fact4=sqrt(F(:,1)'*F(:,1)*Ia*d);
fact5=sqrt(F(:,2)'*F(:,2)*Ia*d);
fact6=sqrt(F(:,3)'*F(:,3)*Ia*d);
wt(:,:)=F(:,:); %/fact4;
mt=[wt(:,1) wt(:,2) wt(:,3)]'*[wt(:,1) wt(:,2) wt(:,3)]*Ia*d; %cek
apakah identitas
%
Gmass=sum(mu*d*f'*f);
GTmass=sum(Ia*d*F'*F);
%generalized
z=zeros(p,3);
%modal
modal=[wb(:,1) wb(:,2) wb(:,3) z(:,1) z(:,2) z(:,3);z(:,1) z(:,2)
z(:,3) wt(:,1) wt(:,2) wt(:,3)];
%
Salf=mu*d*cgea*b;          % momen statik tiap elemen

```

```

%
%VALIDASI ROGER

y1=-0.3;
y2=-0.6;
y3=-1.1;
y4=-1.6;

w=[1 1;1 1];      %weighting factor
Ac=zeros(7,7);
b1=zeros(7,1);
b2=zeros(7,1);
b3=zeros(7,1);
b4=zeros(7,1);

%kr=0.068693:0.00055:0.123198;
%kr=[0.0001 0.0005 0.001 0.004 0.008 0.01 0.03 0.05 0.08 0.1 0.2 0.3
0.4 0.6 0.8 1 2 3 4 5]
kr=0.1:0.01:1
for m=1:length(kr);
    %m
    k=kr(m);
    H1=besselh(1,2,k);
    H0=besselh(0,2,k);
    Ck=H1/(H1+i*H0);      % Theodorsen's function
    Lh(m)=1-2*i*Ck/k;      % Lift - plunge
    La(m)=0.5-2*i*(0.5+(1-i/k)*Ck)/k; % Lift - pitch
    Mh(m)=0.5;            % Moment - plunge
    Ma(m)=3/8-i/k;        % Moment - pitch

    Ac=[1 0 -1*(k^2) ((k^2)./(k^2+y1^2)) (k^2)./(k^2+y2^2)
k^2./(k^2+y3^2) k^2./(k^2+y4^2); 0 k 0 k*y1./(k^2+y1^2)
k*y2./(k^2+y2^2) k*y3./(k^2+y3^2) k*y4./(k^2+y4.^2)]'*[1 0 -1*(k^2)
((k^2)./(k^2+y1^2)) (k^2)./(k^2+y2^2) k^2./(k^2+y3^2) k^2./(k^2+y4^2);
0 k 0 k*y1./(k^2+y1^2) k*y2./(k^2+y2^2) k*y3./(k^2+y3^2)
k*y4./(k^2+y4.^2)]+Ac;
    b1=[1 0 -1*(k^2) ((k^2)./(k^2+y1^2)) (k^2)./(k^2+y2^2)
k^2./(k^2+y3^2) k^2./(k^2+y4^2); 0 k 0 k*y1./(k^2+y1^2)
k*y2./(k^2+y2^2) k*y3./(k^2+y3^2)
k*y4./(k^2+y4.^2)]'*[real(Lh(m)); imag(Lh(m))]+b1;
    b2=[1 0 -1*(k^2) ((k^2)./(k^2+y1^2)) (k^2)./(k^2+y2^2)
k^2./(k^2+y3^2) k^2./(k^2+y4^2); 0 k 0 k*y1./(k^2+y1^2)
k*y2./(k^2+y2^2) k*y3./(k^2+y3^2)
k*y4./(k^2+y4.^2)]'*[real(La(m)); imag(La(m))]+b2;
    b3=[1 0 -1*(k^2) ((k^2)./(k^2+y1^2)) (k^2)./(k^2+y2^2)
k^2./(k^2+y3^2) k^2./(k^2+y4^2); 0 k 0 k*y1./(k^2+y1^2)
k*y2./(k^2+y2^2) k*y3./(k^2+y3^2)
k*y4./(k^2+y4.^2)]'*[real(Mh(m)); imag(Mh(m))]+b3;
    b4=[1 0 -1*(k^2) ((k^2)./(k^2+y1^2)) (k^2)./(k^2+y2^2)
k^2./(k^2+y3^2) k^2./(k^2+y4^2); 0 k 0 k*y1./(k^2+y1^2)
k*y2./(k^2+y2^2) k*y3./(k^2+y3^2)
k*y4./(k^2+y4.^2)]'*[real(Ma(m)); imag(Ma(m))]+b4;

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```

end
%stop
%koefisien nilai roger
  A_Lhr=Ac\b1;
  dA_Lhr1=A_Lhr(1,:);
  dA_Lhr2=A_Lhr(2,:);
  dA_Lhr3=A_Lhr(3,:);
  dA_Lhr4=A_Lhr(4,:);
  dA_Lhr5=A_Lhr(5,:);
  dA_Lhr6=A_Lhr(6,:);
  dA_Lhr7=A_Lhr(7,:);
  A_Lar=Ac\b2;
  dA_Lar1=A_Lar(1,:);
  dA_Lar2=A_Lar(2,:);
  dA_Lar3=A_Lar(3,:);
  dA_Lar4=A_Lar(4,:);
  dA_Lar5=A_Lar(5,:);
  dA_Lar6=A_Lar(6,:);
  dA_Lar7=A_Lar(7,:);
  A_Mhr=Ac\b3;
  dA_Mhr1=A_Mhr(1,:);
  dA_Mhr2=A_Mhr(2,:);
  dA_Mhr3=A_Mhr(3,:);
  dA_Mhr4=A_Mhr(4,:);
  dA_Mhr5=A_Mhr(5,:);
  dA_Mhr6=A_Mhr(6,:);
  dA_Mhr7=A_Mhr(7,:);
  A_Mar=Ac\b4;
  dA_Mar1=A_Mar(1,:);
  dA_Mar2=A_Mar(2,:);
  dA_Mar3=A_Mar(3,:);
  dA_Mar4=A_Mar(4,:);
  dA_Mar5=A_Mar(5,:);
  dA_Mar6=A_Mar(6,:);
  dA_Mar7=A_Mar(7,:);

%nilai roger 4 lag
%for k=1:length(m)      % Run entire blade for range of values k at
blade tip
  %real roger
  Lh_roger_real=1*dA_Lhr1-
  (kr.^2)*dA_Lhr3+((kr.^2)./(kr.^2+y1^2))*dA_Lhr4+(kr.^2)./(kr.^2+y2^2)*
  dA_Lhr5+kr.^2./(kr.^2+y3^2)*dA_Lhr6+kr.^2./(kr.^2+y4^2)*dA_Lhr7;
  La_roger_real=1*dA_Lar1-
  (kr.^2)*dA_Lar3+((kr.^2)./(kr.^2+y1^2))*dA_Lar4+(kr.^2)./(kr.^2+y2^2)*
  dA_Lar5+kr.^2./(kr.^2+y3^2)*dA_Lar6+kr.^2./(kr.^2+y4^2)*dA_Lar7;
  Mh_roger_real=1*dA_Mhr1-
  (kr.^2)*dA_Mhr3+((kr.^2)./(kr.^2+y1^2))*dA_Mhr4+(kr.^2)./(kr.^2+y2^2)*
  dA_Mhr5+kr.^2./(kr.^2+y3^2)*dA_Mhr6+kr.^2./(kr.^2+y4^2)*dA_Mhr7;
  Ma_roger_real=1*dA_Mar1-
  (kr.^2)*dA_Mar3+((kr.^2)./(kr.^2+y1^2))*dA_Mar4+(kr.^2)./(kr.^2+y2^2)*
  dA_Mar5+kr.^2./(kr.^2+y3^2)*dA_Mar6+kr.^2./(kr.^2+y4^2)*dA_Mar7;

  %imaginer roger

  Lh_roger_imag=0*dA_Lhr1+k*dA_Lhr2+0*dA_Lhr3+kr.*y1./(kr.^2+y1^2)*dA_Lh

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r4+kr.*y2./(kr.^2+y2^2)*dA_Lhr5+kr.*y3./(kr.^2+y3^2)*dA_Lhr6+kr.*y4./(
kr.^2+y4.^2)*dA_Lhr7;

La_roger_imag=0*dA_Lar1+k*dA_Lar2+0*dA_Lar3+kr.*y1./(kr.^2+y1^2)*dA_La
r4+kr.*y2./(kr.^2+y2^2)*dA_Lar5+kr.*y3./(kr.^2+y3^2)*dA_Lar6+kr.*y4./(
kr.^2+y4.^2)*dA_Lar7;

Mh_roger_imag=0*dA_Mhr1+k*dA_Mhr2+0*dA_Mhr3+kr.*y1./(kr.^2+y1^2)*dA_Mh
r4+kr.*y2./(kr.^2+y2^2)*dA_Mhr5+kr.*y3./(kr.^2+y3^2)*dA_Mhr6+kr.*y4./(
kr.^2+y4.^2)*dA_Mhr7;

Ma_roger_imag=0*dA_Mar1+k*dA_Mar2+0*dA_Mar3+kr.*y1./(kr.^2+y1^2)*dA_Ma
r4+kr.*y2./(kr.^2+y2^2)*dA_Mar5+kr.*y3./(kr.^2+y3^2)*dA_Mar6+kr.*y4./(
kr.^2+y4.^2)*dA_Mar7;
%end

%=====
=====
%PLOT ROGER
%=====
=====

figure(4)
subplot(2,1,1)
title('Real Lh')
plot(kr,Lh_roger_real,kr,real(Lh))
legend('Lh roger','Lh Theodorsen')
%axis([min(kr),max(kr),0.75,1.05])
xlabel('reduced frequency, k');ylabel('Lh');
grid on
%stop
    subplot(2,1,2)
    title('imaginer Lh')
    plot(kr,Lh_roger_imag,kr,imag(Lh))
    %plot(kr,imag(Lh))
    %axis([min(kr),max(kr),-1.2,0])
    legend('Lh roger','Lh Theodorsen')
    xlabel('reduced frequency, k');ylabel('Lh');
    grid on
%stop
figure(5)
subplot(2,1,1)
title('Real La')
plot(kr,La_roger_real,kr,real(La))
%axis([min(kr),max(kr),-0.8,0.6])
legend('La roger','La Theodorsen')
xlabel('reduced frequency, k');ylabel('La');
grid on
    subplot(2,1,2)
    title('Imaginer La')
    plot(kr,La_roger_imag,kr,imag(La))
    %axis([min(kr),max(kr),-2,0])
    legend('La roger','La Theodorsen')
    xlabel('reduced frequency, k');ylabel('La');
    grid on

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```

figure(6)
subplot(2,1,1)
title('Real Mh')
plot(kr,Mh_roger_real,kr,real(Mh))
%axis([min(kr),max(kr),0,1])
legend('Mh roger','Mh Theodorsen')
xlabel('reduced frequency, k');ylabel('Mh');
grid on
    subplot(2,1,2)
    title('Imaginer Mh')
    plot(kr,Mh_roger_imag,kr,imag(Mh))
    %axis([min(kr),max(kr),0,1])
    legend('Mh roger','Mh Theodorsen')
    xlabel('reduced frequency, k');ylabel('Mh');
    grid on

figure(7)
subplot(2,1,1)
title('Real Ma')
plot(kr,Ma_roger_real,kr,real(Ma))
legend('Ma roger','Ma Theodorsen')
%axis([min(kr),max(kr),0,0.5])
xlabel('reduced frequency, k');ylabel('Ma');
grid on
    subplot(2,1,2)
    title('Imaginer Ma')
    plot(kr,Ma_roger_imag,kr,imag(Ma))
    %axis([min(kr),max(kr),-1,0])
    legend('Ma roger','Ma Theodorsen')
    xlabel('reduced frequency, k');ylabel('Ma');
    grid on

for n=1:3                                % Calculate static moments for
mode coupling
    for m=1:3
        Sar(n,m)=sum(Salf*wb(:,n).*wt(:,m))/Gmass(1,n);
        Iaf(n,m)=sum(Ia*d*wt(:,n).*wt(:,m));
        Mma(n,m)=sum(mu*d*wb(:,n).*wb(:,m));
        Saf(n,m)=sum(Salf*wt(:,n).*wb(:,m))/GTmass(1,n);
    end
end
%
Identi=[1 0 0;
        0 1 0;
        0 0 1];
genmass=[Identi -Sar; -Saf Identi];
%stop
%genmass=[Mma Sar; Saf Iaf];
%
nlag=3;
nstate=3+nlag;
y1=0.2;
y2=0.4;
y3=0.6;
y4=0.8;
%
```

```

%mencaari theodorsen
%
m=0.068693:0.00055:0.123198;
%
%roger untuk run program Aer
%
Vfwd=linspace(mn,mx,300); % Set range of airspeeds for helicopter
%
ww=ones(7*p,7*p);
coefi=zeros(p,7,7);
rhlas=zeros(p,7);
rhlhs=zeros(p,7);
rhmas=zeros(p,7);
rhms=zeros(p,7);
%
for q=1:length(Vfwd)
    for t=1:p
        Vbaru=(Vfwd(q)+O*R*mid(p+1-t));
        kt=wntp(1)*b/Vbaru;
        H1=besselh(1,2,kt);
        H0=besselh(0,2,kt);
        Ck=H1/(H1+i*H0); % Theodorsen's function
        Lh=(1-2*i*Ck/kt); % Lift - plunge
        La=(0.5-2*i*(0.5+(1-i/kt)*Ck)/kt); % Lift - pitch
        Mh=(0.5); % Moment - plunge
        Ma=(3/8-i/kt); % Moment - pitch
        %
        Lhh= 2*pi*kt^2*d*Lh;
        Laa= 2*pi*kt^2*d*(La-(0.5+a)*Lh)*b;
        Mhh= 2*pi*kt^2*d*(Mh-(0.5+a)*Lh)*b;
        Maa= 2*pi*kt^2*d*(Ma-(0.5+a)*(La+Mh)+(0.5+a)^2*Lh)*b^2;
        %
        b11=[1 0 -1*(kt^2) ((kt^2)/(kt^2+y1^2)) (kt^2)/(kt^2+y2^2)
kt^2/(kt^2+y3^2) kt^2./(kt^2+y4^2);
0 kt 0 kt*y1/(kt^2+y1^2) kt*y2/(kt^2+y2^2)
kt*y3/(kt^2+y3^2) kt*y4/(kt^2+y4^2)];
        %
        bb=b11'*b11;
        %
        coef(t, :, :)=bb(:, :);
        %
        btlh=b11'*[real(Lhh) imag(Lhh)]';
        btla=b11'*[real(Laa) imag(Laa)]';
        btmh=b11'*[real(Mhh) imag(Mhh)]';
        btma=b11'*[real(Maa) imag(Maa)]';
        %
        rhlh(t, :)=btlh(:);
        rhla(t, :)=btla(:);
        rhma(t, :)=btma(:);
        rhmh(t, :)=btmh(:);
    end
end
%pause
coefi=coefi+coef;
rhlhs=rhlhs+rhlh;
rhlas=rhlas+rhla;
rhms=rhms+rhmh;

```



```

        rhmas=rhmas+rhma;
end
%
for j=1:p
    bb(:,:)=coefi(j, :, :);
    rlh(:,1)=rhlhs(j, :);
    rla(:,1)=rhlas(j, :);
    rma(:,1)=rhmhs(j, :);
    rmh(:,1)=rhmhs(j, :);
%
    ALh=inv(bb)*rlh;
    ALa=inv(bb)*rla;
    AMh=inv(bb)*rmh;
    AMa=inv(bb)*rma;
%
    AALh(j, :)=ALh(:);
    AALa(j, :)=ALa(:);
    AAMh(j, :)=AMh(:);
    AAMa(j, :)=AMa(:);
end
%
%state space
%
Identity=[1 0 0 0 0 0;
          0 1 0 0 0 0;
          0 0 1 0 0 0;
          0 0 0 1 0 0;
          0 0 0 0 1 0;
          0 0 0 0 0 1];
%
for n=1:3
    for m=1:3
        A2(n,m)=0.5*rho*sum(AALh(:,3).*wb(:,n).*wb(:,m))/Gmass(1,n);
        A2(n,3+m)=0.5*rho*sum(AALa(:,3).*wb(:,n).*wt(:,m))/Gmass(1,n);

A2(3+n,m)=0.5*rho*sum(AAMh(:,3).*wt(:,n).*wb(:,m))/GTmass(1,n);

A2(3+n,3+m)=0.5*rho*sum(AAMh(:,3).*wt(:,n).*wt(:,m))/GTmass(1,n);
    end
end
%
Msp=genmass-b^2*A2;
%
Kec=linspace(1,1000);
%
fb=wb'*wb;
ft=wt'*wt;
%
for tr=1:length(Kec)
    vel=Kec(tr);
    for j=1:p
        Vf(j)=(vel+(O*R*mid(p+1-t)));
    end
    for n=1:3
        for m=1:3

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```

A0(n,m)=0.5*rho*sum(Vf(:)'.^2*AALh(:,1).*wb(:,n).*wb(:,m))/Gmass(1,n);
A0(n,3+m)=0.5*rho*sum(Vf(:)'.^2*AALa(:,1).*wb(:,n).*wt(:,m))/Gmass(1,n);
A0(3+n,m)=0.5*rho*sum(Vf(:)'.^2*AAMh(:,1).*wt(:,n).*wb(:,m))/GTmass(1,n);
A0(3+n,3+m)=0.5*rho*sum(Vf(:)'.^2*AAMh(:,1).*wt(:,n).*wt(:,m))/GTmass(1,n);
%
A1(n,m)=0.5*rho*sum(Vf(:).*AALh(:,2).*wb(:,n).*wb(:,m))/Gmass(1,n);
A1(n,3+m)=0.5*rho*sum(Vf(:).*AALa(:,2).*wb(:,n).*wt(:,m))/Gmass(1,n);
A1(3+n,m)=0.5*rho*sum(Vf(:).*AAMh(:,2).*wt(:,n).*wb(:,m))/GTmass(1,n);
A1(3+n,3+m)=0.5*rho*sum(Vf(:).*AAMh(:,2).*wt(:,n).*wt(:,m))/GTmass(1,n);
%
A3(n,m)=0.5*rho*sum(Vf(:)'.^2*AALh(:,4).*wb(:,n).*wb(:,m))/Gmass(1,n);
A3(n,3+m)=0.5*rho*sum(Vf(:)'.^2*AALa(:,4).*wb(:,n).*wt(:,m))/Gmass(1,n);
A3(3+n,m)=0.5*rho*sum(Vf(:)'.^2*AAMh(:,4).*wt(:,n).*wb(:,m))/GTmass(1,n);
A3(3+n,3+m)=0.5*rho*sum(Vf(:)'.^2*AAMh(:,4).*wt(:,n).*wt(:,m))/GTmass(1,n);
%
A4(n,m)=0.5*rho*sum(Vf(:)'.^2*AALh(:,5).*wb(:,n).*wb(:,m))/Gmass(1,n);
A4(n,3+m)=0.5*rho*sum(Vf(:)'.^2*AALa(:,5).*wb(:,n).*wt(:,m))/Gmass(1,n);
A4(3+n,m)=0.5*rho*sum(Vf(:)'.^2*AAMh(:,5).*wt(:,n).*wb(:,m))/GTmass(1,n);
A4(3+n,3+m)=0.5*rho*sum(Vf(:)'.^2*AAMh(:,5).*wt(:,n).*wt(:,m))/GTmass(1,n);
%
A5(n,m)=0.5*rho*sum(Vf(:)'.^2*AALh(:,6).*wb(:,n).*wb(:,m))/Gmass(1,n);
A5(n,3+m)=0.5*rho*sum(Vf(:)'.^2*AALa(:,6).*wb(:,n).*wt(:,m))/Gmass(1,n);
A5(3+n,m)=0.5*rho*sum(Vf(:)'.^2*AAMh(:,6).*wt(:,n).*wb(:,m))/GTmass(1,n);

```

```

A5(3+n,3+m)=0.5*rho*sum(Vf(:)'.^2*AAMh(:,6).*wt(:,n).*wt(:,m))/GTmass(
1,n);
%
A6(n,m)=0.5*rho*sum(Vf(:)'.^2*AALh(:,7).*wb(:,n).*wb(:,m))/Gmass(1,n);
A6(n,3+m)=0.5*rho*sum(Vf(:)'.^2*AALa(:,7).*wb(:,n).*wt(:,m))/Gmass(1,n
);
A6(3+n,m)=0.5*rho*sum(Vf(:)'.^2*AAMh(:,7).*wt(:,n).*wb(:,m))/GTmass(1,
n);
A6(3+n,3+m)=0.5*rho*sum(Vf(:)'.^2*AAMh(:,7).*wt(:,n).*wt(:,m))/GTmass(
1,n);
end
end
%
Dsp=-b*A1;
Ksp=[wnbp(1).^2 0 0 0 0 0;
0 wnbp(2).^2 0 0 0 0;
0 0 wnbp(3).^2 0 0 0;
0 0 0 wntp(1).^2 0 0;
0 0 0 0 wntp(2).^2 0;
0 0 0 0 0 wntp(3).^2]-(A0);
%
%matriks baru
%
Ivev=zeros(6,6);
for n=1:3 % Calculate static moments
for mode coupling
Ivev(n,n)=sum(Vf(:).*wb(:,n).*wb(:,n))/fb(n,n);
Ivev(3+n,3+n)=sum(Vf(:).*wt(:,n).*wt(:,n))/ft(n,n);
end
%
MatrikSP=[zeros(6,6) Identity zeros(6,6)
zeros(6,6) zeros(6,6) zeros(6,6);
-1*(inv(Msp)*Ksp) -1*(inv(Msp)*Dsp) inv(Msp)*A3
inv(Msp)*A4 inv(Msp)*A5 inv(Msp)*A6;
zeros(6,6) Identity -Ivev*y1/b
zeros(6,6) zeros(6,6) zeros(6,6);
zeros(6,6) Identity zeros(6,6) -
Ivev*y2/b zeros(6,6) zeros(6,6);
zeros(6,6) Identity zeros(6,6)
zeros(6,6) -Ivev*y3/b zeros(6,6);
zeros(6,6) Identity zeros(6,6)
zeros(6,6) zeros(6,6) -Ivev*y4/b];
%
% menghitung eigenvalues
%
Z(tr,:)=eig(MatrikSP(:,:),'nobalance');
[Zz(tr,:),index]=sort(imag(Z(tr,:)));
Z1(tr,:)=Z(tr,index);
ZR(tr,:)=real(Z1(tr,:));
ZI(tr,:)=imag(Z1(tr,:));
for jj=1:6

```

```

        freqi(tr,jj)=ZI(tr,30+jj);
        dampi(tr,jj)=-
ZR(tr,30+jj)/sqrt(ZI(tr,30+jj)^2+ZR(tr,30+jj)^2);
    end
    freq(tr,1)=freqi(tr,1);
    freq(tr,2)=freqi(tr,2);
    freq(tr,3)=freqi(tr,4);
    damp(tr,1)=dampi(tr,1);
    damp(tr,2)=dampi(tr,2);
    damp(tr,3)=dampi(tr,4);
    %Z1'
    %ZI'
    %
    %pause
end
%

%stop
%ZR
%pause
Vf'
Kec'
freq
damp
%pause
%
for tr=1:1 :length(Kec)
    Vef(tr)=(Kec(tr)+(O*R))./12;
end
%size(MatrikSP)
%size(eig(MatrikSP))
%
%=====
=====
%plot Vg dan Vf
%=====
=====
%
%for tr=1:1 :length(Kec)
%    Kec(tr), freq(tr,:)
%end
ktip=b*wntp(1)./(Vfwd+O*R);          % Define reduced frequency for tip
x=1./ktip;
if mode==3
    V2=(freq(tr,2)*b/12).*x; % velocity corresponding to actual oTR(1)
in fwd flight
else
    V2=(freq(tr,4)*b/12).*x; % velocity corresponding to actual oTR(1)
in fwd flight
end

figure(8)
subplot(2,1,1)
plot(Vef,freq)
grid on
xlabel('Velocity(ft/s)'); ylabel('freq');

```

```

%legend('bending 1','bending 2','bending 3','tors1 1','tors1 2','tors1
3')
%axis([0,50,50,400])
subplot(2,1,2)
plot(Vef,damp)
grid on
xlabel('Velocity (ft/s)'); ylabel('damp');
%legend('bending 1','bending 2','bending 3','tors1 1','tors1 2','tors1
3')
%axis([0,50,-400,0])

figure(9)
plot(-ZR,ZI)
grid on
xlabel('real'); ylabel('imag');
legend('bending 1','bending 2','bending 3','tors1 1','tors1 2','tors1
3')
axis([-100,100,-900,900])
%
vel=0;
%
for j=1:p
    Vf(j)=(vel+(O*R*mid(p+1-t)));
end
for n=1:3
    for m=1:3

A0(n,m)=0.5*rho*sum(Vf(:)'.^2*AALh(:,1).*wb(:,n).*wb(:,m))/Gmass(1,n);

A0(n,3+m)=0.5*rho*sum(Vf(:)'.^2*AALa(:,1).*wb(:,n).*wt(:,m))/Gmass(1,n
);

A0(3+n,m)=0.5*rho*sum(Vf(:)'.^2*AAMh(:,1).*wt(:,n).*wb(:,m))/GTmass(1,
n);

A0(3+n,3+m)=0.5*rho*sum(Vf(:)'.^2*AAMh(:,1).*wt(:,n).*wt(:,m))/GTmass(
1,n);
        %

A1(n,m)=0.5*rho*sum(Vf(:).*AALh(:,2).*wb(:,n).*wb(:,m))/Gmass(1,n);

A1(n,3+m)=0.5*rho*sum(Vf(:).*AALa(:,2).*wb(:,n).*wt(:,m))/Gmass(1,n);

A1(3+n,m)=0.5*rho*sum(Vf(:).*AAMh(:,2).*wt(:,n).*wb(:,m))/GTmass(1,n);

A1(3+n,3+m)=0.5*rho*sum(Vf(:).*AAMh(:,2).*wt(:,n).*wt(:,m))/GTmass(1,n
);
        %

A3(n,m)=0.5*rho*sum(Vf(:)'.^2*AALh(:,4).*wb(:,n).*wb(:,m))/Gmass(1,n);

A3(n,3+m)=0.5*rho*sum(Vf(:)'.^2*AALa(:,4).*wb(:,n).*wt(:,m))/Gmass(1,n
);

A3(3+n,m)=0.5*rho*sum(Vf(:)'.^2*AAMh(:,4).*wt(:,n).*wb(:,m))/GTmass(1,
n);

```

```

A3(3+n,3+m)=0.5*rho*sum(Vf(:)'.^2*AAMh(:,4).*wt(:,n).*wt(:,m))/GTmass(
1,n);
%
A4(n,m)=0.5*rho*sum(Vf(:)'.^2*AALh(:,5).*wb(:,n).*wb(:,m))/Gmass(1,n);
A4(n,3+m)=0.5*rho*sum(Vf(:)'.^2*AALa(:,5).*wb(:,n).*wt(:,m))/Gmass(1,n
);
A4(3+n,m)=0.5*rho*sum(Vf(:)'.^2*AAMh(:,5).*wt(:,n).*wb(:,m))/GTmass(1,
n);
A4(3+n,3+m)=0.5*rho*sum(Vf(:)'.^2*AAMh(:,5).*wt(:,n).*wt(:,m))/GTmass(
1,n);
%
A5(n,m)=0.5*rho*sum(Vf(:)'.^2*AALh(:,6).*wb(:,n).*wb(:,m))/Gmass(1,n);
A5(n,3+m)=0.5*rho*sum(Vf(:)'.^2*AALa(:,6).*wb(:,n).*wt(:,m))/Gmass(1,n
);
A5(3+n,m)=0.5*rho*sum(Vf(:)'.^2*AAMh(:,6).*wt(:,n).*wb(:,m))/GTmass(1,
n);
A5(3+n,3+m)=0.5*rho*sum(Vf(:)'.^2*AAMh(:,6).*wt(:,n).*wt(:,m))/GTmass(
1,n);
%
A6(n,m)=0.5*rho*sum(Vf(:)'.^2*AALh(:,7).*wb(:,n).*wb(:,m))/Gmass(1,n);
A6(n,3+m)=0.5*rho*sum(Vf(:)'.^2*AALa(:,7).*wb(:,n).*wt(:,m))/Gmass(1,n
);
A6(3+n,m)=0.5*rho*sum(Vf(:)'.^2*AAMh(:,7).*wt(:,n).*wb(:,m))/GTmass(1,
n);
A6(3+n,3+m)=0.5*rho*sum(Vf(:)'.^2*AAMh(:,7).*wt(:,n).*wt(:,m))/GTmass(
1,n);
end
end
%
Dsp=-b*A1;
Ksp=[wnbp(1).^2 0 0 0 0 0;
0 wnbp(2).^2 0 0 0 0;
0 0 wnbp(3).^2 0 0 0;
0 0 0 wntp(1).^2 0 0;
0 0 0 0 wntp(2).^2 0;
0 0 0 0 0 wntp(3).^2]-(A0);
%
%matriks baru
%
Ivev=zeros(6,6);
for n=1:3 % Calculate static moments for
mode coupling
Ivev(n,n)=sum(Vf(:).*wb(:,n).*wb(:,n))/fb(n,n);

```

```

        Ivec(3+n,3+n)=sum(Vf(:).*wt(:,n).*wt(:,n))/ft(n,n);
end
%
MatrikSP=[zeros(6,6)          Identity      zeros(6,6)  zeros(6,6)
zeros(6,6)  zeros(6,6);
          -1*(inv(Msp)*Ksp) -1*(inv(Msp)*Dsp) inv(Msp)*A3  inv(Msp)*A4
inv(Msp)*A5  inv(Msp)*A6;
          zeros(6,6)          Identity      -Ivec*y1/b  zeros(6,6)
zeros(6,6)  zeros(6,6);
          zeros(6,6)          Identity      zeros(6,6)  -Ivec*y2/b
zeros(6,6)  zeros(6,6);
          zeros(6,6)          Identity      zeros(6,6)  zeros(6,6)
-Ivec*y3/b  zeros(6,6);
          zeros(6,6)          Identity      zeros(6,6)  zeros(6,6)
zeros(6,6)  -Ivec*y4/b];
%
for k=1:36
    xsp0(k)=0;
end
xsp0(1)=0.5;
t=0:0.01:2;
C=[1 zeros(1,35)];
sys = ss(MatrikSP, [], C, []);
%initial(sys,xsp0)
[y,t,x] = initial(sys,xsp0,t)
%initialplot(sys,xsp0,t)
%hh(1)=0;
%hh
figure(10)
%subplot(2,1,1)
plot(t,y)
grid on
%subplot(2,1,2)
%plot(t,dispp(50,t))
%grid on

```